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Generalized Coupling Management in Complex Engineering Systems Optimization

Decomposition-based design optimization strategies are used to solve complex engineering system problems that might be otherwise unsolvable. Yet, the associated computational cost can be prohibitively high due to the often large number of iterations needed for coordination of subproblem solutions. To reduce this cost one may exploit the fact that some systems may be weakly coupled and their interactions can be suspended with little loss in solution accuracy. Suspending such interactions is usually based on the analyst's experience or experimental observation. This article introduces an explicit measure of coupling strength among interconnected subproblems in a decomposed system optimization problem, along with a systematic way for calculating it. The strength measure is then used to suspend weak couplings and, thus, improve system solution strategies such as the model coordination method. Examples show that the resulting strategy may decrease the number of required function evaluations significantly.

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1 Introduction

Analysis and design of complex engineering systems often require decomposing the problem into smaller subsystems in order to handle comprehension and computation difficulties. System solution is obtained through coordination of subsystem solutions. Such coordination is strongly affected by the interconnections or coupling of the subsystems. Intuitively, totally “uncoupled” subsystems would require the simplest possible coordination and “fully coupled” systems would gain little by decomposition and would defy coordination. A number of different coupling measures have been proposed, (e.g., [1–3]), each with its own range of applicability, advantages, and disadvantages. The exact definition of coupling to be used depends on the nature of the system problem at hand, and in our case this is the solution of system design optimization problems.

In the multidisciplinary design optimization (MDO) community, coordination and coupling information are often represented by the, design structure matrix (DSM) developed by Steward [4], which assumes that all task relations have strengths of one or zero (exist or not exist). Gebala and Eppinger [5] proposed a nonbinary DSM that utilizes problem dependent information to assign numerical values to the couplings reflecting the strength of the relationship between tasks and the overall design. This approach utilizes engineering judgment to set the numerical values, based in part on the perceived consequences of having to estimate information that is not yet known. While it can be useful in setting the order of design tasks, it is not readily extensible to design optimization, since that would require coding engineering judgment into the algorithm and changing the order of design tasks as the optimization is carried out. Wagner [6,7] introduced the functional dependence table (FDT), also referred to as the design incidence matrix, to assist in model-based decomposition of optimization problems; see also Krishnamachari and Papalambros [8] and Michelena and Papalambros [9]. The FDT is essentially the Jacobian matrix of problem

functions and, as such, it does not contain binary values. However, partitioning the FDT requires filtering the partial derivative (or “sensitivity” values) to a zero–one representation. Moreover, Jacobian values are different at different points in the design space so a universal coupling strength cannot be established unequivocally. Interestingly, the DSM is the adjacency matrix of the FDT.

Closer to our present approach, Sobieszczanski-Sobieski [10] investigated the effect of design variable changes on the interaction variables in an internally coupled system. He used the chain rule to relate total derivatives of system outputs with respect to system design variables to local system derivatives. These total derivatives are found by solving the resulting set of equations termed the global sensitivity equations (GSE). Subsequently, English and Bloebaum [2,11] proposed a method that utilizes the total derivative-based coupling sensitivity analysis to suspend interaction variables between systems during MDO coordination cycles. Emphasis was placed on single-level MDO approaches, such as multidisciplinary feasible methods [2,11]. The GSE formulations in these research efforts did not include optimality conditions in the definition of coupling, and so the design variables are assumed to be independent of each other. The work presented in this article augments the definition of GSE to include variable coupling implied by the need to satisfy optimality. The idea of generalizing the GSE to include optimality was first proposed by Sobieszczanski-Sobieski [12] as a numerical tool in a bilevel decomposition strategy. In contrast, the generalization of the GSE in this article is used in the context of coupling strength quantification. A key difference between the present approach and that of Sobieski is the provision for local and global copies of design and interaction variables. This provision was not included in Ref. [12]; however, it is necessary to include these local and global copies of the design variables in the calculation of coupling with a suspension strategy.

Several other authors have examined strength-based coupling between systems in decomposed optimization problems [13,14]. Reyer et al. [15] and Fathy et al. [3] proposed the use of optimality conditions to characterize coupling. They defined coupling relative to the solution method, for example, by comparing the optimality conditions of a sequential solution strategy with the optimality conditions of the undecomposed system optimization problem. However, the effect of interaction variables (interconnections) between

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systems on the coupling measure was not directly considered in that work. Moreover, Fathy's work was motivated by integration of design and control problems, and so it examined coupling between only two subsystems and assumed some objective function separability. Thus, the effect of all system variables on coupling strength was not fully explored. The relationship between various proposed coupling measures for design and control problems was established in Refs. [16,17]. Recently, Peters et al. [18] studied the relationship between Fathy's coupling vector and the controllability Gramian matrix in design and control problems.

Further work exploring the effect of system variables on coupling strength was performed by Allison et al. [19,20]. Allison's work utilized coupling in the decomposition process, with the intention of decomposing a complex design problem in an optimal way. Coupling information was also used by Han and Papalambros to formulate a suspension strategy for analytical target cascading (ATC) [21]. Han's work on complex system optimization also incorporated uncertainty, as did the work by Han and Papalambros and Gu et al. on collaborative optimization [22,23].

This paper examines how the GSEs can be modified to account for satisfaction of optimality conditions in defining subsystem coupling relevant to system optimization. This modification leads to a measure of coupling strength that can be used to suspend weak variable linking in an MDO strategy. The strategy used for demonstration is Kirsch's model coordination method [24], chosen for its feasible intermediate solutions and easy applicability to design problems. Section 2 poses the problem under consideration. Section 3 shows how the new coupling measure can be derived accounting for satisfaction of optimality. Section 4 presents the model coordination method with variable suspension strategy. Section 5 presents a mathematical example with the model coordination method, and Secs. 6 and 7 present example implementations for a simple structure and a direct current (DC) motor, respectively. The paper concludes in Sec. 8 with a discussion of limitations and future work.

2 The System Optimization Problem

We consider design optimization of a supersystem that has been decomposed into four coupled systems, Fig. 1. Each system is assumed to perform its own optimization problem using its own analysis models and information from the other systems. A general nonhierarchical structure is assumed, with a hierarchical decomposition being a special case. Each system interacts (is "coupled") potentially with all the other systems though the interaction variables $\mathbf{y}_{ij} \in \mathbb{R}^{l_{ij}}$, where l_{ij} is the dimension of \mathbf{y}_{ij} . A direction of information flow is implied, so \mathbf{y}_{ij} represents information going from system i to system j , and \mathbf{y}_{ji} is used to represent information from analysis models within system i used to compute the functions (objectives or constraints) in the system optimization problem. This allows representation of various MDO schemes that involve information exchange among analyses as well as design decisions.

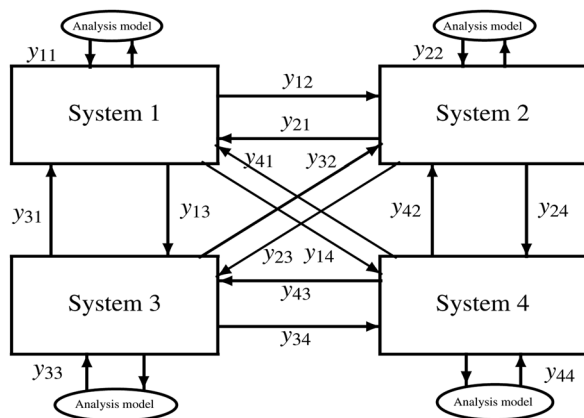


Fig. 1 Nonhierarchical system interactions notation

The design optimization problem for a supersystem with objective $F : \mathbb{R}^{\sum_{i=1}^N n_i} \rightarrow \mathbb{R}$ to be decomposed in N systems is stated as follows:

$$\begin{aligned} \min_{\{\mathbf{x}_i | i=1, \dots, N\}} & F(f_1(\mathbf{x}_1, \dots, \mathbf{x}_N, \mathbf{y}_{11}, \dots, \mathbf{y}_{N1}), \dots \\ & \dots, f_N(\mathbf{x}_1, \dots, \mathbf{x}_N, \mathbf{y}_{1N}, \dots, \mathbf{y}_{NN}), \mathbf{x}_1, \dots, \mathbf{x}_N) \\ \text{subject to} & \mathbf{g}_i(\mathbf{x}_1, \dots, \mathbf{x}_N, \mathbf{y}_{1i}, \dots, \mathbf{y}_{Ni}) \leq \mathbf{0} \\ & \mathbf{h}_i(\mathbf{x}_1, \dots, \mathbf{x}_N, \mathbf{y}_{1i}, \dots, \mathbf{y}_{Ni}) = \mathbf{0} \\ & i = 1, \dots, N, \quad j = 1, \dots, N \end{aligned} \quad (1a)$$

where

$$\mathbf{y}_{ij} = Y_{ij}(\mathbf{x}_1, \dots, \mathbf{x}_N, \mathbf{y}_{1i}, \dots, \mathbf{y}_{Ni}) \quad (1b)$$

represents the analysis models, namely, the functional dependence of the interaction variables on the design variables of each system $\mathbf{x}_i \in \mathbb{R}^{n_i}$ and on all other interaction variables. The vector \mathbf{x}_i includes local variables specific to system i and shared variables that are common in at least two systems. Hence, the \mathbf{x}_i 's may have common components. The system objectives $f_i : \mathbb{R}^{q_i} \rightarrow \mathbb{R}$ and constraints $\mathbf{g}_i : \mathbb{R}^{q_i} \rightarrow \mathbb{R}^{m_i}$, $\mathbf{h}_i : \mathbb{R}^{q_i} \rightarrow \mathbb{R}^{o_i}$ may depend on other systems' design variables $\mathbf{x}_1, \dots, \mathbf{x}_N$, as well as on the interaction variables that bring information from each of the other systems. Also, n_i, m_i, o_i, l_{ij} are the dimensions of the vectors $\mathbf{x}_i, \mathbf{g}_i, \mathbf{h}_i, \mathbf{y}_{ij}$, respectively, and $q_i \triangleq \sum_{j=1}^N (n_j + l_{ji})$ is the total number of design and interaction variables associated with system i . Note that in the problem statement above no assumption is made on the form of the decomposition or the structure of the objective and constraint functions.

After decomposition, the design optimization problem for system i is stated as follows:

$$\begin{aligned} \min_{\mathbf{x}_i} & f_i(\mathbf{x}_1, \dots, \mathbf{x}_N, \mathbf{y}_{1i}, \dots, \mathbf{y}_{Ni}) \\ \text{subject to} & \mathbf{g}_i(\mathbf{x}_1, \dots, \mathbf{x}_N, \mathbf{y}_{1i}, \dots, \mathbf{y}_{Ni}) \leq \mathbf{0} \\ & \mathbf{h}_i(\mathbf{x}_1, \dots, \mathbf{x}_N, \mathbf{y}_{1i}, \dots, \mathbf{y}_{Ni}) = \mathbf{0} \end{aligned} \quad (2)$$

We assume that the optimal solution of the N system optimization problems in Eq. (2) combined with the analysis equations in Eq. (1b) will yield the supersystem optimization problem optimal solution in Eq. (1a). If the objective function F is separable, i.e., $F(f_1, f_2, \dots, f_N) = \sum_{i=1}^N F_i(f_i)$, and $\partial F_i / \partial f_i > 0 \forall i$, then this assumption will be valid, as shown for the case with two objective functions in Ref. [17]. Other problem formulations, not requiring separability, may also satisfy this assumption. Convergence to the supersystem optimal solution depends on the decomposition strategy utilized.

A variable suspension strategy during a solution process will ignore the link between two systems for one or more iterations k . Namely, during suspension we set

$$\mathbf{x}_{i,k+1} = \mathbf{x}_{i,k}, \quad \mathbf{y}_{ij,k+1} = \mathbf{y}_{ij,k} \quad (3)$$

where the index after the comma indicates iteration number.

3 Generalized Coupling Strength and the Modified Global Sensitivity Equations

Sobieski's GSEs [10] provide a foundation for the work presented later in this article. Following the notation of Fig. 1, consider a three-system problem with the following analysis equations

$$\mathbf{y}_{12} = Y_{12}(\mathbf{x}, \mathbf{y}_{23}, \mathbf{y}_{31}) \quad (4a)$$

$$\mathbf{y}_{23} = Y_{23}(\mathbf{x}, \mathbf{y}_{12}, \mathbf{y}_{31}) \quad (4b)$$

$$\mathbf{y}_{31} = Y_{31}(\mathbf{x}, \mathbf{y}_{12}, \mathbf{y}_{23}) \quad (4c)$$

where \mathbf{x} are the supersystem variables. In the original Sobieski derivation all system outputs are considered identical, and so

$\mathbf{y}_{12} = \mathbf{y}_{13}$, $\mathbf{y}_{21} = \mathbf{y}_{23}$, $\mathbf{y}_{31} = \mathbf{y}_{32}$; further, no internal simulation/analysis is assumed, and so $\mathbf{y}_{11} = \mathbf{y}_{22} = \mathbf{y}_{33} = \mathbf{0}$.

Using the chain rule, the total derivatives of system outputs with respect to system design variables, $d\mathbf{y}_{12}/d\mathbf{x}$, $d\mathbf{y}_{23}/d\mathbf{x}$, and $d\mathbf{y}_{31}/d\mathbf{x}$, are given by the GSEs [10]

$$\begin{bmatrix} I & -\frac{\partial \mathbf{y}_{12}}{\partial \mathbf{y}_{23}} & -\frac{\partial \mathbf{y}_{12}}{\partial \mathbf{y}_{31}} \\ -\frac{\partial \mathbf{y}_{23}}{\partial \mathbf{y}_{12}} & I & -\frac{\partial \mathbf{y}_{23}}{\partial \mathbf{y}_{31}} \\ -\frac{\partial \mathbf{y}_{31}}{\partial \mathbf{y}_{12}} & -\frac{\partial \mathbf{y}_{31}}{\partial \mathbf{y}_{23}} & I \end{bmatrix} \begin{bmatrix} \frac{d\mathbf{y}_{12}}{d\mathbf{x}} \\ \frac{d\mathbf{y}_{23}}{d\mathbf{x}} \\ \frac{d\mathbf{y}_{31}}{d\mathbf{x}} \end{bmatrix} = \begin{bmatrix} \frac{\partial \mathbf{y}_{12}}{\partial \mathbf{x}} \\ \frac{\partial \mathbf{y}_{23}}{\partial \mathbf{x}} \\ \frac{\partial \mathbf{y}_{31}}{\partial \mathbf{x}} \end{bmatrix} \quad (5)$$

The left-hand side matrix in Eq. (5) contains the partial derivatives (sensitivities) of system outputs with respect to changes in other systems' output. The right-hand side matrix contains the partial derivatives of system outputs with respect to system design variables. These derivatives are evaluated analytically or numerically.

We now proceed to develop a modification of the GSEs to account for optimality of supersystem design. The Lagrangians of the N problems in Eq. (2) are

$$L_i = f_i(\mathbf{x}_1, \dots, \mathbf{x}_N, \mathbf{y}_{1i}, \dots, \mathbf{y}_{Ni}) + \boldsymbol{\mu}_i^T \mathbf{g}_i(\mathbf{x}_1, \dots, \mathbf{x}_N, \mathbf{y}_{1i}, \dots, \mathbf{y}_{Ni}) + \boldsymbol{\lambda}_i^T \mathbf{h}_i(\mathbf{x}_1, \dots, \mathbf{y}_{1i}, \dots, \mathbf{y}_{Ni}) \quad i = 1, \dots, N \quad (6)$$

where $\boldsymbol{\lambda}_i$, $\boldsymbol{\mu}_i$ are the Lagrange multipliers for the equality, inequality constraints, respectively, and the vector of supersystem variables \mathbf{x} is partitioned into a set of vectors $\mathbf{x}_1, \mathbf{x}_2, \dots, \mathbf{x}_N$. The first order Karush-Kuhn-Tucker (KKT) stationarity conditions [25] are written as

$$\begin{aligned} \frac{\partial f_i}{\partial \mathbf{x}_i} + \boldsymbol{\mu}_i^T \frac{\partial \mathbf{g}_i}{\partial \mathbf{x}_i} + \boldsymbol{\lambda}_i^T \frac{\partial \mathbf{h}_i}{\partial \mathbf{x}_i} + \sum_{j=1}^N \frac{\partial f_i}{\partial \mathbf{y}_{ji}} \frac{d\mathbf{y}_{ji}}{d\mathbf{x}_i} \\ + \sum_{j=1}^N \boldsymbol{\mu}_j^T \left(\frac{\partial \mathbf{g}_j}{\partial \mathbf{y}_{ji}} \frac{d\mathbf{y}_{ji}}{d\mathbf{x}_i} \right) + \sum_{j=1}^N \boldsymbol{\lambda}_j^T \left(\frac{\partial \mathbf{h}_j}{\partial \mathbf{y}_{ji}} \frac{d\mathbf{y}_{ji}}{d\mathbf{x}_i} \right) = \mathbf{0}^T \\ \boldsymbol{\lambda}_i \neq \mathbf{0}, \quad \boldsymbol{\mu}_i \geq \mathbf{0}, \quad \boldsymbol{\mu}_i^T \mathbf{g}_i = 0 \quad i = 1, \dots, N \end{aligned} \quad (7)$$

The total derivatives $d\mathbf{y}_{ij}/d\mathbf{x}_i$ in Eq. (7) can be found by taking the derivatives of Eq. (1b)

$$\frac{d\mathbf{y}_{jp}}{d\mathbf{x}_i} = \frac{\partial \mathbf{y}_{jp}}{\partial \mathbf{x}_i} + \frac{\partial \mathbf{y}_{jp}}{\partial \mathbf{y}_{1j}} \frac{d\mathbf{y}_{1j}}{d\mathbf{x}_i} + \dots + \frac{\partial \mathbf{y}_{jp}}{\partial \mathbf{y}_{Nj}} \frac{d\mathbf{y}_{Nj}}{d\mathbf{x}_i} \quad i = 1, \dots, N, \quad j = 1, \dots, N, \quad p = 1, \dots, N \quad (8)$$

collected in matrix form as

$$\begin{bmatrix} I & -\frac{\partial \mathbf{y}_{12}}{\partial \mathbf{y}_{21}} & \dots & -\frac{\partial \mathbf{y}_{12}}{\partial \mathbf{y}_{NN}} \\ -\frac{\partial \mathbf{y}_{21}}{\partial \mathbf{y}_{12}} & I & & -\frac{\partial \mathbf{y}_{21}}{\partial \mathbf{y}_{NN}} \\ \vdots & & \ddots & \\ -\frac{\partial \mathbf{y}_{NN}}{\partial \mathbf{y}_{12}} & -\frac{\partial \mathbf{y}_{NN}}{\partial \mathbf{y}_{21}} & & I \end{bmatrix} \begin{bmatrix} \frac{d\mathbf{y}_{12}}{d\mathbf{x}_1} \dots \frac{d\mathbf{y}_{12}}{d\mathbf{x}_N} \\ \frac{d\mathbf{y}_{21}}{d\mathbf{x}_1} \dots \frac{d\mathbf{y}_{21}}{d\mathbf{x}_N} \\ \vdots \\ \frac{d\mathbf{y}_{NN}}{d\mathbf{x}_1} \dots \frac{d\mathbf{y}_{NN}}{d\mathbf{x}_N} \end{bmatrix} = \begin{bmatrix} \frac{\partial \mathbf{y}_{12}}{\partial \mathbf{x}_1} \dots \frac{\partial \mathbf{y}_{12}}{\partial \mathbf{x}_N} \\ \frac{\partial \mathbf{y}_{21}}{\partial \mathbf{x}_1} \dots \frac{\partial \mathbf{y}_{21}}{\partial \mathbf{x}_N} \\ \vdots \\ \frac{\partial \mathbf{y}_{NN}}{\partial \mathbf{x}_1} \dots \frac{\partial \mathbf{y}_{NN}}{\partial \mathbf{x}_N} \end{bmatrix} \quad (9)$$

which are Sobieski's GSEs [12]. These must be extended to account for the presence of local and global copies of variables and thus provide a measure of coupling strength when systems are suspended, as we will see next.

Suppose that the link of system i with the rest of the systems is weak and the system i variables can be suspended. When system i

is suspended, it is not re-optimized in response to changes in the optimal values of other systems. Then the system design variables \mathbf{x}_i become parameters in the new optimization problem with \mathbf{x}_i suspended. The coupling strength is defined as $dF^*(\mathbf{x}_i)/d\mathbf{x}_i$, namely, the sensitivity of the supersystem optimal objective with respect to \mathbf{x}_i . When this sensitivity is small, changes in \mathbf{x}_i will result in only small changes to the supersystem optimum, and thus it is a useful measure of coupling. From the implicit function theorem the conditions in Eq. (7) can be solved for each \mathbf{x}_i , except for the optimality condition corresponding to suspended system i

$$\hat{\mathbf{y}}_{jp} = Y_{jp}(\hat{\mathbf{x}}_1, \dots, \hat{\mathbf{x}}_N, \hat{\mathbf{y}}_{1j}, \dots, \hat{\mathbf{y}}_{Nj}) \quad (10a)$$

$$\hat{\mathbf{x}}_l = X_l(\hat{\mathbf{x}}_1, \dots, \hat{\mathbf{x}}_{l-1}, \hat{\mathbf{x}}_{l+1}, \dots, \hat{\mathbf{x}}_N, \hat{\mathbf{y}}_{1l}, \dots, \hat{\mathbf{y}}_{Nl}) \quad l = 1, \dots, N, \quad j = 1, \dots, N, \quad p = 1, \dots, N \quad l \neq i \quad (10b)$$

Here X_l is the corresponding solution function. The analysis equations in Eq. (1b) are also rewritten with "hats": The system design variables $\hat{\mathbf{x}}_i$ and the interaction variables $\hat{\mathbf{y}}_{ij}$ will have different sensitivities from \mathbf{x}_i and \mathbf{y}_{ij} , respectively, hence the hat. Note that the optimality condition corresponding to system i with \mathbf{x}_i suspended is not included in Eq. (10b). Moreover, the hat is not used on the suspended variable \mathbf{x}_i .

Let us now consider how this coupling strength is computed. The sensitivity of the supersystem optimal objective in Eq. (1a) with respect to \mathbf{x}_i can be expressed as

$$\Gamma_i \triangleq \frac{dF}{d\mathbf{x}_i} = \sum_{p=1}^N \sum_{j=1}^N \left(\frac{\partial F}{\partial f_p} \frac{\partial f_p}{\partial \mathbf{x}_j} \frac{d\hat{\mathbf{x}}_j}{d\mathbf{x}_i} \right) + \sum_{p=1}^N \sum_{j=1}^N \left(\frac{\partial F}{\partial f_p} \frac{\partial f_p}{\partial \mathbf{y}_{jp}} \frac{d\hat{\mathbf{y}}_{jp}}{d\mathbf{x}_i} \right) + \sum_{j=1}^N \frac{\partial F}{\partial \mathbf{x}_j} \frac{d\hat{\mathbf{x}}_j}{d\mathbf{x}_i} \quad (11)$$

The $1 \times n_i$ vector Γ_i , which is the partial derivative of the system objective function with respect to the variables of system i , is the coupling function for system \mathbf{x}_i . In matrix form Eq. (11) is written as

$$\Gamma \triangleq \begin{bmatrix} \Gamma_1 \\ \Gamma_2 \\ \vdots \\ \Gamma_N \end{bmatrix}^T = \begin{bmatrix} \frac{\partial F}{\partial f_1} \frac{\partial f_1}{\partial \mathbf{y}_{11}} \\ \vdots \\ \frac{\partial F}{\partial f_N} \frac{\partial f_N}{\partial \mathbf{y}_{1N}} \\ \frac{\partial F}{\partial f_1} \frac{\partial f_1}{\partial \mathbf{y}_{21}} \\ \vdots \\ \frac{\partial F}{\partial f_N} \frac{\partial f_N}{\partial \mathbf{y}_{2N}} \\ \vdots \\ \frac{\partial F}{\partial f_1} \frac{\partial f_1}{\partial \mathbf{y}_{N1}} \\ \vdots \\ \frac{\partial F}{\partial f_N} \frac{\partial f_N}{\partial \mathbf{y}_{NN}} \end{bmatrix}^T \begin{bmatrix} \frac{d\hat{\mathbf{y}}_{11}}{d\mathbf{x}_1} & \frac{d\hat{\mathbf{y}}_{11}}{d\mathbf{x}_2} & \dots & \frac{d\hat{\mathbf{y}}_{11}}{d\mathbf{x}_N} \\ \frac{d\hat{\mathbf{y}}_{12}}{d\mathbf{x}_1} & \frac{d\hat{\mathbf{y}}_{12}}{d\mathbf{x}_2} & \dots & \frac{d\hat{\mathbf{y}}_{12}}{d\mathbf{x}_N} \\ \vdots & \vdots & \ddots & \vdots \\ \frac{d\hat{\mathbf{y}}_{NN}}{d\mathbf{x}_1} & \frac{d\hat{\mathbf{y}}_{NN}}{d\mathbf{x}_2} & \dots & \frac{d\hat{\mathbf{y}}_{NN}}{d\mathbf{x}_N} \end{bmatrix} + \begin{bmatrix} \sum_{j=1}^N \frac{\partial F}{\partial \mathbf{x}_j} \frac{d\hat{\mathbf{x}}_j}{d\mathbf{x}_1} \\ \vdots \\ \sum_{j=1}^N \frac{\partial F}{\partial \mathbf{x}_j} \frac{d\hat{\mathbf{x}}_j}{d\mathbf{x}_N} \end{bmatrix}^T + \begin{bmatrix} \sum_{p=1}^N \sum_{j=1}^N \left(\frac{\partial F}{\partial f_p} \frac{\partial f_p}{\partial \mathbf{x}_j} \frac{d\hat{\mathbf{x}}_j}{d\mathbf{x}_1} \right) \\ \vdots \\ \sum_{p=1}^N \sum_{j=1}^N \left(\frac{\partial F}{\partial f_p} \frac{\partial f_p}{\partial \mathbf{x}_j} \frac{d\hat{\mathbf{x}}_j}{d\mathbf{x}_N} \right) \end{bmatrix}^T \quad (12)$$

where Γ is defined as the vector collection of all Γ_i 's. Note that Γ is a row vector with dimensionality $1 \times \sum_{i=1}^N n_i$; the four matrices

on the right side of the equation have dimensionality $1 \times N^3$, $N^3 \times \sum_{i=1}^N n_i$, $1 \times \sum_{i=1}^N n_i$, and $1 \times \sum_{i=1}^N n_i$, respectively.

Since some analysis functions do not yield meaningful solutions at an infeasible point, it will be assumed that a feasible design point can be found. Given such a feasible point, most of the elements of the coupling function in Eq. (11) can be evaluated readily. The objective partial derivatives are evaluated first, analytically or numerically. The next two quantities to find are the total derivatives $d\hat{x}_j/dx_i$ and $d\hat{y}_{ij}/dx_i$. Equation (10a) can be used to determine $d\hat{x}_j/dx_i$, $d\hat{y}_{ij}/dx_i$ based on local partial derivatives as follows:

$$\begin{aligned} \frac{d\hat{y}_{jp}}{dx_i} = & \frac{\partial \hat{y}_{jp}}{\partial \hat{x}_1} \frac{d\hat{x}_1}{dx_i} + \dots + \frac{\partial \hat{y}_{jp}}{\partial \hat{x}_N} \frac{d\hat{x}_N}{dx_i} + \frac{\partial \hat{y}_{jp}}{\partial \hat{y}_{1j}} \frac{d\hat{y}_{1j}}{dx_i} + \dots \\ & + \frac{\partial \hat{y}_{jp}}{\partial \hat{y}_{Nj}} \frac{d\hat{y}_{Nj}}{dx_i} \quad i = 1, \dots, N, \quad j = 1, \dots, N, \quad p = 1, \dots, N \end{aligned} \quad (13)$$

Suspending system 1 and taking the derivatives of Eq. (10b) we get

$$\begin{aligned} \frac{d\hat{x}_j}{dx_i} = & \frac{\partial \hat{x}_j}{\partial \hat{x}_1} \frac{d\hat{x}_1}{dx_i} + \dots + \frac{\partial \hat{x}_j}{\partial \hat{x}_N} \frac{d\hat{x}_N}{dx_i} \\ & + \frac{\partial \hat{x}_j}{\partial \hat{y}_{1j}} \frac{d\hat{y}_{1j}}{dx_i} + \dots + \frac{\partial \hat{x}_j}{\partial \hat{y}_{Nj}} \frac{d\hat{y}_{Nj}}{dx_i} \\ & i = 2, \dots, N, \quad j = 1, \dots, N, \quad p = 1, \dots, N \end{aligned} \quad (14)$$

Collecting the resulting equations in matrix form gives the modified global sensitivity equations (MGSE),

$$\begin{bmatrix} I & \dots & -\frac{\partial \hat{y}_{12}}{\partial \hat{y}_{NN}} & -\frac{\partial \hat{y}_{12}}{\partial \hat{x}_1} & \dots & -\frac{\partial \hat{y}_{12}}{\partial \hat{x}_N} \\ \frac{\partial \hat{y}_{21}}{\partial \hat{y}_{12}} & \dots & I & -\frac{\partial \hat{y}_{21}}{\partial \hat{x}_1} & \dots & -\frac{\partial \hat{y}_{21}}{\partial \hat{x}_N} \\ \vdots & \vdots & \vdots & \vdots & \ddots & \vdots \\ -\frac{\partial \hat{y}_{NN}}{\partial \hat{y}_{12}} & \dots & -\frac{\partial \hat{y}_{NN}}{\partial \hat{y}_{21}} & -\frac{\partial \hat{y}_{NN}}{\partial \hat{x}_1} & \dots & -\frac{\partial \hat{y}_{NN}}{\partial \hat{x}_2} \\ \frac{\partial \hat{x}_2}{\partial \hat{y}_{12}} & \dots & \frac{\partial \hat{x}_2}{\partial \hat{y}_{21}} & I & \dots & \frac{\partial \hat{x}_2}{\partial \hat{x}_3} \\ \vdots & \vdots & \vdots & \vdots & \ddots & \vdots \\ -\frac{\partial \hat{x}_N}{\partial \hat{y}_{12}} & \dots & -\frac{\partial \hat{x}_N}{\partial \hat{y}_{21}} & -\frac{\partial \hat{x}_N}{\partial \hat{x}_2} & \dots & I \end{bmatrix} \begin{bmatrix} \frac{d\hat{y}_{12}}{dx_i} \\ \vdots \\ \frac{d\hat{y}_{NN}}{dx_i} \\ \frac{d\hat{x}_2}{dx_i} \\ \vdots \\ \frac{d\hat{x}_N}{dx_i} \end{bmatrix} = \begin{bmatrix} \frac{\partial \hat{y}_{12}}{\partial \hat{x}_1} \\ \vdots \\ \frac{\partial \hat{y}_{N1}}{\partial \hat{x}_N} \\ \frac{\partial \hat{x}_2}{\partial \hat{x}_1} \\ \vdots \\ \frac{\partial \hat{x}_N}{\partial \hat{x}_1} \end{bmatrix} \quad (15)$$

The local derivatives $\partial \hat{y}_{ij}/\partial \hat{x}_i$, $\partial \hat{y}_{ij}/\partial \hat{y}_{ij}$, $\partial \hat{x}_j/\partial \hat{x}_i$, $\partial \hat{x}_j/\partial \hat{y}_{ij}$ can be computed either analytically or numerically using finite differences. The terms $\partial \hat{y}_{ij}/\partial \hat{x}_i$, $\partial \hat{y}_{ij}/\partial \hat{y}_{ij}$ represent local analysis derivatives, while $\partial \hat{x}_j/\partial \hat{x}_i$, $\partial \hat{x}_j/\partial \hat{y}_{ij}$ represent derivatives of the optimum with respect to parameters. We use the KKT conditions at the optimum to predict the local derivatives [26], based on the assumption that constraints at the optimum remain active as \hat{x}_1 is changed. Those constraints which are inactive do not effect the local derivatives, and may therefore be neglected in this calculation. Second order derivatives of the objective and active constraints are required, as well as the Lagrange multipliers associated with the optimum design.

The MGSEs are different from the original GSEs in that they include the optimality conditions as part of the coupled system of equations used to compute the solution sensitivity. Thus, the MGSEs account for the relationship between optimization and analysis. The key point here is that in a decomposed supersystem, the effect of one system on another may be small at nonoptimal feasible points but large at the optimum, which, after all, is the point of interest.

4 The Model Coordination Method With Variable Suspension

We consider now the model coordination method of Kirsch and Schoeffler [24,27] along with a variable suspension strategy during optimization. Suspension reduces the number of system optimizations yielding a more efficient strategy for solving the model coordination method. Note that, while suspension is being demonstrated here for the model coordination method, the suspension strategy can be applied to decomposition-based methods involving partitioning and coordination of subsystems. Such methods may be either hierarchical or nonhierarchical.

Model coordination is a hierarchical two-level method that solves independent system optimization problems by fixing their coordination variable. The convergence of the model coordination method is not guaranteed [24]. However, the method remains attractive in design problems because even if convergence of the coordination is not attained, the intermediate solutions are feasible and usually represent an improvement in the objective function. Consequently, the method is also known as the feasible decomposition method.

Consider a supersystem decomposed using the model coordination method into several coupled systems. The undecomposed supersystem optimization problem is

$$\begin{aligned} \min_{\mathbf{z}, \mathbf{v}} F(\mathbf{z}, \mathbf{v}) \\ \text{subject to } \mathbf{g}(\mathbf{z}, \mathbf{v}) \leq \mathbf{0}, \quad \mathbf{h}(\mathbf{z}, \mathbf{v}) = \mathbf{0} \end{aligned} \quad (16)$$

where $\mathbf{z} \in \mathbb{R}^n$ is the vector of design variables, and $\mathbf{v} \in \mathbb{R}^{n_3}$ is the vector of coordination variables. Let \mathbf{z} be partitioned into $\mathbf{z} = [\mathbf{z}_1^T, \mathbf{z}_2^T]^T$, $n = n_1 + n_2$, and assume that the problem and its objective function can be decomposed into the following two systems

$$\begin{aligned} F(\mathbf{z}, \mathbf{v}) = f_1(\mathbf{z}_1, \mathbf{v}) + f_2(\mathbf{z}_2, \mathbf{v}) \\ \mathbf{g}_i(\mathbf{z}_i, \mathbf{v}) \leq \mathbf{0}, \quad \mathbf{h}_i(\mathbf{z}_i, \mathbf{v}) = \mathbf{0} \quad i = 1, 2 \end{aligned} \quad (17)$$

where $\mathbf{z}_i \in \mathbb{R}^{n_i}$ is the vector of design variables. The model coordination method converts the supersystem optimization problem in Eq. (16) into the decomposed two-level problem in Eq. (18) and shown in Fig. 2, by fixing the coordination variable \mathbf{v} .

$$\begin{aligned} \text{UpperLevel : } \min_{\mathbf{v}} f_1(\mathbf{z}_1, \mathbf{v}) + f_2(\mathbf{z}_2, \mathbf{v}) \\ \text{LowerLevel : } \min_{\mathbf{z}_i} f_i(\mathbf{z}_i, \mathbf{v}) \\ \mathbf{g}_i(\mathbf{z}_i, \mathbf{v}) \leq \mathbf{0}, \quad \mathbf{h}_i(\mathbf{z}_i, \mathbf{v}) = \mathbf{0} \quad i = 1, 2 \end{aligned} \quad (18)$$

To characterize coupling associated with the model coordination method the problem formulation in Eq. (18) is written in terms of the coupled systems notation of Eq. (1a). Letting $\mathbf{x}_1 = \mathbf{z}_1$, $\mathbf{x}_2 = \mathbf{z}_2$, $\mathbf{x}_3 = \mathbf{v}$, Eq. (18) can be rewritten as

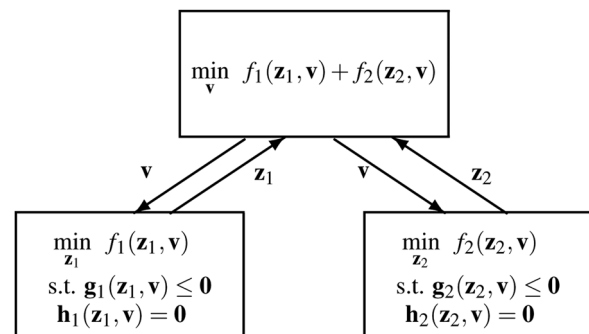


Fig. 2 The model coordination method

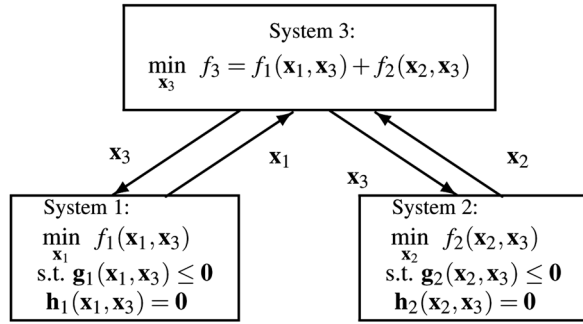


Fig. 3 The model coordination method written in the coupled systems optimization framework

$$\begin{aligned}
 \text{System 1:} \quad & \min_{\mathbf{x}_1} f_1(\mathbf{x}_1, \mathbf{x}_3) \\
 & \mathbf{g}_1(\mathbf{x}_1, \mathbf{x}_3) \leq \mathbf{0}, \quad \mathbf{h}_1(\mathbf{x}_1, \mathbf{x}_3) = \mathbf{0} \\
 \text{System 2:} \quad & \min_{\mathbf{x}_2} f_2(\mathbf{x}_2, \mathbf{x}_3) \\
 & \mathbf{g}_2(\mathbf{x}_2, \mathbf{x}_3) \leq \mathbf{0}, \quad \mathbf{h}_2(\mathbf{x}_2, \mathbf{x}_3) = \mathbf{0} \\
 \text{System 3:} \quad & \min_{\mathbf{x}_3} f_3 = f_1(\mathbf{x}_3, \mathbf{x}_2) + f_2(\mathbf{x}_3, \mathbf{x}_3)
 \end{aligned} \tag{19}$$

where system 3 represents the upper-level coordinator, Fig. 3.

Let us now consider a hierarchical coupling suspension (HCS) strategy, an optimization strategy that intelligently suspends a system's optimization variables in the case of weak coupling, during at least some of the iterations. Suspending variable \mathbf{x}_1 of system 1 results in "systems with suspension" and an isolated system, as shown in Fig. 4.

The HCS strategy flowchart is shown in Fig. 5. The algorithm estimates $dx_2^*(\mathbf{x}_1)/d\mathbf{x}_1$, $dx_3^*(\mathbf{x}_1)/d\mathbf{x}_1$, which are the sensitivities of the optimal solution with respect to suspended variable \mathbf{x}_1 . The algorithm then computes $df_3^*(\mathbf{x}_1)/d\mathbf{x}_1$, the sensitivity of system 3 optimal objective with respect to \mathbf{x}_1 , which indicates coupling strength. This strength is used to determine whether to continue to suspend \mathbf{x}_1 . After suspension, system design variable changes resulting from the optimization process alter the sensitivities $dx_2^*(\mathbf{x}_1)/d\mathbf{x}_1$, $dx_3^*(\mathbf{x}_1)/d\mathbf{x}_1$, $df_3^*(\mathbf{x}_1)/d\mathbf{x}_1$ and require computing new sensitivities. A trust region criterion can be used to avoid frequent updating of derivatives. After convergence, the optimization problem is solved without suspension to validate the suspension decision.

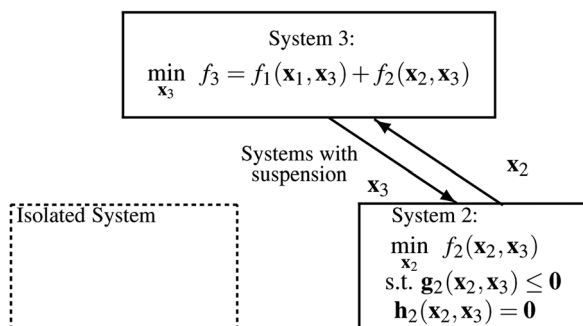


Fig. 4 The model coordination method with variable \mathbf{x}_1 suspended

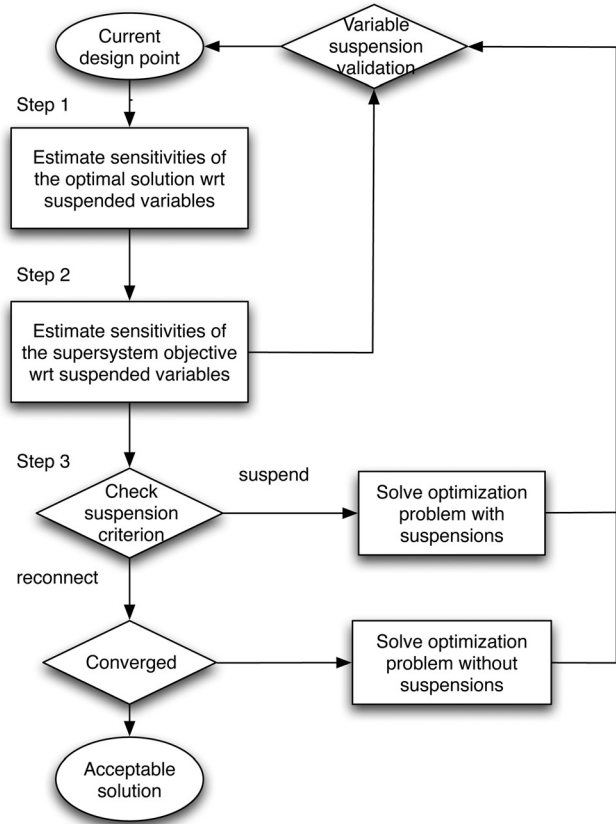


Fig. 5 The HCS algorithm flowchart

The algorithm's steps are described in more detail as follows:

Step 0: Initialize. Set $k=0$ with an initial feasible design $\mathbf{x}_{1,0}$, $\mathbf{x}_{2,0}$, $\mathbf{x}_{3,0}$. Set $k=1$ and optimize systems 1 and 2 to yield $\mathbf{x}_{1,1}^*$ and $\mathbf{x}_{2,1}^*$, respectively; complete the iteration by utilizing $\mathbf{x}_{1,1}^*$ and $\mathbf{x}_{2,1}^*$ to optimize system 3.

Step 1: Estimate sensitivities of the optimal solution with respect to suspended variables. The sensitivities $dx_2^*(\mathbf{x}_1)/d\mathbf{x}_1$, $dx_3^*(\mathbf{x}_1)/d\mathbf{x}_1$ are needed to determine the coupling strength $df_3^*(\mathbf{x}_1)/d\mathbf{x}_1$. Note that \mathbf{x}_1 is a parameter in the system with suspension, so these sensitivities are parameter sensitivities at the optimal solution of systems 2 and 3. To this end, represent systems 2 and 3 in the format of Eq. (10b)

$$(\text{System 2}) \quad \hat{\mathbf{x}}_2 = X_2(\mathbf{x}_1, \hat{\mathbf{x}}_3) \tag{20}$$

$$(\text{System 3}) \quad \hat{\mathbf{x}}_3 = X_3(\mathbf{x}_1, \hat{\mathbf{x}}_2) \tag{21}$$

where the X_i 's are the corresponding solution functions. The system design variable $\hat{\mathbf{x}}_i$ will have different sensitivities from \mathbf{x}_i , hence the hat. The derivatives of $\hat{\mathbf{x}}_2$ and $\hat{\mathbf{x}}_3$ with respect to \mathbf{x}_1 in Eqs. (20) and (21) can be expressed from Eq. (15) as

$$\begin{bmatrix} I & -\frac{\partial \hat{\mathbf{x}}_2}{\partial \hat{\mathbf{x}}_3} \\ -\frac{\partial \hat{\mathbf{x}}_3}{\partial \hat{\mathbf{x}}_2} & I \end{bmatrix} \begin{bmatrix} \frac{d\hat{\mathbf{x}}_2}{d\mathbf{x}_1} \\ \frac{d\hat{\mathbf{x}}_3}{d\mathbf{x}_1} \end{bmatrix} = \begin{bmatrix} \frac{\partial \hat{\mathbf{x}}_2}{\partial \mathbf{x}_1} \\ \frac{\partial \hat{\mathbf{x}}_3}{\partial \mathbf{x}_1} \end{bmatrix} \tag{22}$$

The optimal solution sensitivities $dx_2^*(\mathbf{x}_1)/d\mathbf{x}_1$, $dx_3^*(\mathbf{x}_1)/d\mathbf{x}_1$ can be determined by solving the linear equations in Eq. (22) given the local derivatives $\partial \hat{\mathbf{x}}_2 / \partial \hat{\mathbf{x}}_3$, $\partial \hat{\mathbf{x}}_3 / \partial \hat{\mathbf{x}}_2$, $\partial \hat{\mathbf{x}}_2 / \partial \mathbf{x}_1$, $\partial \hat{\mathbf{x}}_3 / \partial \mathbf{x}_1$. A solution to these equations can generally be found; due to the presence of the identity matrices, it

is unlikely that the matrix of partial derivatives would be rank deficient. The difficulty is that $d\hat{x}_2/dx_1$ and $d\hat{x}_3/dx_1$ have to be calculated at the optimal solution of the problem with x_1 suspended to satisfy the functional relationship they were derived from. This requirement does not provide a criterion for suspension, only a check at the final iteration to see if the suspension was correct. If the feasible solution chosen as a starting point is far from optimality, it is possible that the coupling strength at the optimum will be significantly different from the strength calculated at this point. However, if the method used produces an improvement in the solution to the supersystem problem, then it can be expected that the coupling strength calculated will be more accurate as the solution method progresses.

Here, Eq. (22) is evaluated at the current feasible point and the values of $d\hat{x}_2/dx_1$ and $d\hat{x}_3/dx_1$ are only estimates of the sensitivities at the optimum.

Step 2: Estimate sensitivities of the supersystem objective with respect to suspended variables. The suspension decision depends on the value of $df_3^*(x_1)/dx_1$, which indicates coupling strength. To explain the meaning of the suspension decision associated with $df_3^*(x_1)/dx_1$, assume a first-order Taylor series expansion of the system 3 objective function at the optimum with suspension

$$\partial f_3^* = \frac{\partial f_3^*}{\partial x_1} \partial x_1 + \frac{\partial f_3^*}{\partial x_2} \partial x_2 + \frac{\partial f_3^*}{\partial x_3} \partial x_3 \quad (23)$$

The optimal solution sensitivities $d\hat{x}_2(x_1)/dx_1$, $d\hat{x}_3(x_1)/dx_1$ can be used to relate ∂x_1 to ∂x_2 and ∂x_3 at the optimum with x_1 suspended

$$\partial x_2 = \frac{d\hat{x}_2}{dx_1} \partial x_1, \quad \partial x_3 = \frac{d\hat{x}_3}{dx_1} \partial x_1 \quad (24)$$

Then Eq. (23) gives

$$\partial f_3^* = \Gamma_1 \partial x_1 = \left(\frac{\partial f_3^*}{\partial x_1} + \frac{\partial f_3^*}{\partial x_2} \frac{d\hat{x}_2}{dx_1} + \frac{\partial f_3^*}{\partial x_3} \frac{d\hat{x}_3}{dx_1} \right) \partial x_1 \quad (25)$$

where Γ_1 is the coupling function defined from Eq. (11) as

$$\Gamma_1 \triangleq \frac{df_3^*(x_1)}{dx_1} = \frac{\partial f_3^*}{\partial x_1} + \frac{\partial f_3^*}{\partial x_2} \frac{d\hat{x}_2}{dx_1} + \frac{\partial f_3^*}{\partial x_3} \frac{d\hat{x}_3}{dx_1} \quad (26)$$

The coupling function Γ_1 represents the effect of a perturbation ∂x_1 on the optimal objective function value of system 3 with suspension. If Γ_1 is "small" then one can assume weak coupling and suspend variable x_1 .

Similarly, if x_2 is suspended we get

$$\partial f_3^* = \Gamma_2 \partial x_2 = \left(\frac{\partial f_3^*}{\partial x_1} \frac{d\hat{x}_1}{dx_2} + \frac{\partial f_3^*}{\partial x_2} + \frac{\partial f_3^*}{\partial x_3} \frac{d\hat{x}_3}{dx_2} \right) \partial x_2 \quad (27)$$

$$\Gamma_2 \triangleq \frac{df_3^*(x_2)}{dx_2} = \frac{\partial f_3^*}{\partial x_1} \frac{d\hat{x}_1}{dx_2} + \frac{\partial f_3^*}{\partial x_2} + \frac{\partial f_3^*}{\partial x_3} \frac{d\hat{x}_3}{dx_2} \quad (28)$$

If Γ_2 is small then one can suspend variable x_2 . Similar equations can be derived for problems with a greater number of subsystems, by expanding Eq. (11).

Step 3: Suspension criterion. The suspension criterion uses the relative magnitude of the Γ_i 's. Here, if $\|\Gamma_1\| < c\|\Gamma_2\|$ then suspend x_1 , if $\|\Gamma_2\| < c\|\Gamma_1\|$ then suspend x_2 . The coupling parameter $c \ll 1$ is chosen based on the designer's experi-

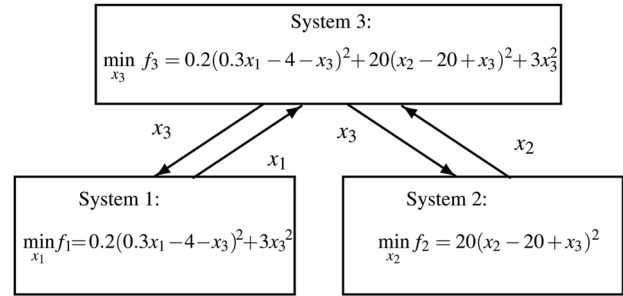


Fig. 6 The model coordination method for the unconstrained optimization example

ence. If Γ_2 is "much larger" than Γ_1 then x_1 can be suspended because it will have relatively little effect on ∂f_3^* . Similarly, if Γ_1 is much larger than Γ_2 then x_2 can be suspended.

Step 4: Suspension Validation. After isolating system 1, the design variables of systems 2 and 3 change during optimization with a corresponding change ∂x_1^* in the estimated x_1^* . A large change in x_1^* can cause large changes in $d\hat{x}_2/dx_1$ and $d\hat{x}_3/dx_1$, making their prediction invalid. If $\|\partial x_1^*\| < \delta$ then estimates are considered valid. The parameter $\delta > 0$ is defined as the radius of the trust region where the linear approximation is considered acceptable. Note that this criterion does not take into consideration other design variables' effect. If $\|\partial x_1^*\| > \delta$ then the estimates are considered invalid and $d\hat{x}_2/dx_1$, $d\hat{x}_3/dx_1$, and Γ_1 must be updated. This update requires performing one system 1 optimization and recomputing $d\hat{x}_2/dx_1$, $d\hat{x}_3/dx_1$, and Γ_1 .

Step 5: Reconnecting the suspended system. If any condition in Steps 3 or 4 is violated, then the isolated system must be reconnected.

Step 6: Termination rules. The algorithm is terminated typically if $\|x_{3,k-1} - x_{3,k}\| < \epsilon$.

The advantage of using the HCS strategy in model coordination is the expected computational savings associated with variable suspension. The computational tradeoff is between reduced system optimization runs and computation of sensitivities.

5 Suspension Strategy Example

The following example is a simple unconstrained optimization problem, which is sufficient to illustrate the key ideas of the HCS strategy and the procedural approach for coupling calculation. The example focuses on two main ideas. The first is to

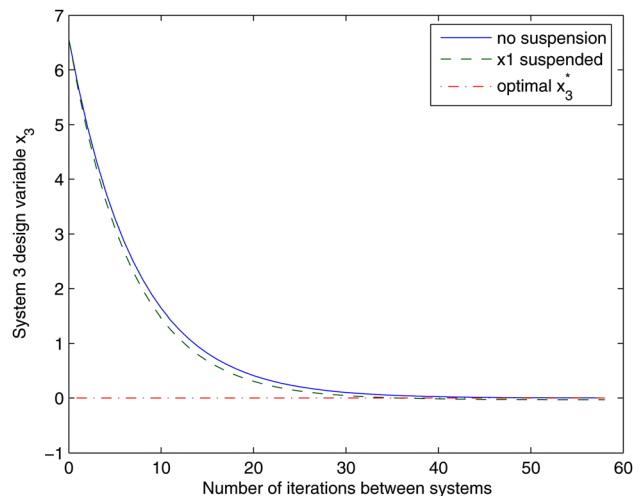


Fig. 7 System iterations with and without suspension

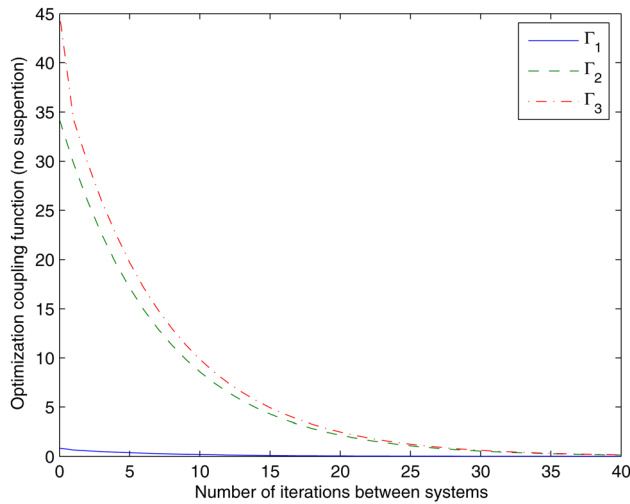


Fig. 8 Optimization coupling function behavior without suspension

demonstrate the HCS strategy on a weakly coupled system. The second is to show the computational savings of the HCS strategy when applied to the model coordination method.

Consider the problem

$$\min_{x_1, x_2, x_3} f_3 = 0.2(0.3x_1 - 4 - x_3)^2 + 20(x_2 - 20 + x_3)^2 + 3x_3^2 \quad (29)$$

The stationarity conditions are

$$\begin{aligned} \frac{\partial f_3}{\partial x_1} &= 0.4(0.3x_1 - 4 - x_3)0.3 = 0 \\ \frac{\partial f_3}{\partial x_2} &= 40(x_2 - 20 + x_3) = 0 \\ \frac{\partial f_3}{\partial x_3} &= -0.4(0.3x_1 - 4 - x_3) + 40(x_2 - 20 + x_3) + 6x_3 = 0 \end{aligned}$$

and the optimal solution is $x_1^* = 4/0.3$, $x_2^* = 20$, and $x_3^* = 0$. The model coordination method is used as shown in Fig. 6. To calculate the coupling function, the optimal solution sensitivities $dx_2^*(x_1)/dx_1$, $dx_3^*(x_1)/dx_1$ must be determined first using Eq. (22),

$$\begin{aligned} \begin{bmatrix} 1 & 1 \\ 0.862 & 1 \end{bmatrix} \begin{bmatrix} \frac{dx_2}{dx_1} \\ \frac{dx_3}{dx_1} \end{bmatrix} &= \begin{bmatrix} 0 \\ 0.0026 \end{bmatrix}, \quad \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} \frac{dx_1}{dx_3} \\ \frac{dx_2}{dx_3} \end{bmatrix} = \begin{bmatrix} 3.33 \\ -1 \end{bmatrix} \\ \begin{bmatrix} 1 & -3.333 \\ -0.0026 & 1 \end{bmatrix} \begin{bmatrix} \frac{dx_1}{dx_2} \\ \frac{dx_3}{dx_2} \end{bmatrix} &= \begin{bmatrix} 0 \\ -0.862 \end{bmatrix} \\ \begin{bmatrix} \frac{dx_2}{dx_1} \\ \frac{dx_3}{dx_1} \end{bmatrix} = \begin{bmatrix} -0.019 \\ 0.019 \end{bmatrix}, \quad \begin{bmatrix} \frac{dx_1}{dx_2} \\ \frac{dx_3}{dx_2} \end{bmatrix} = \begin{bmatrix} -2.899 \\ -0.869 \end{bmatrix}, \quad \begin{bmatrix} \frac{dx_1}{dx_3} \\ \frac{dx_2}{dx_3} \end{bmatrix} = \begin{bmatrix} 3.33 \\ -1 \end{bmatrix} \end{aligned} \quad (30)$$

The coupling functions for systems 1 and 2 are determined from Eq. (26) as

$$\Gamma_1 = \frac{\partial f_3}{\partial x_1} + \frac{\partial f_3}{\partial x_2} \frac{dx_2}{dx_1} + \frac{\partial f_3}{\partial x_3} \frac{dx_3}{dx_1}, \quad \Gamma_2 = \frac{\partial f_3}{\partial x_1} \frac{dx_1}{dx_2} + \frac{\partial f_3}{\partial x_2} + \frac{\partial f_3}{\partial x_3} \frac{dx_3}{dx_2} \quad (31)$$

Applying the HCS strategy, set the initial design $x_{1,0} = 11.619$, $x_{2,0} = 12.381$ and optimize system 3 to get $x_{3,0} = 6.5637$.

The coupling functions in the first iteration are $\Gamma_1 = 0.738$, $\Gamma_2 = 34.2$. Hence, system 1 is isolated and x_1 is suspended. The remaining systems 2 and 3 are optimized until termination. In the course of this optimization, the coupling strength for system 1 is not calculated at each iteration; since the coupling strength for system 2 is so much greater, it is assumed that system 1 will be suspended for the majority of the algorithm. Figure 7 compares iterations with and without suspension. Notice that suspension of x_1 has very little effect on the optimal solution of the overall system. As a result, ignoring system 1 reduces the computational time by a third and yields the same optimal solution. Figure 8 compares the coupling functions for the no suspension case versus

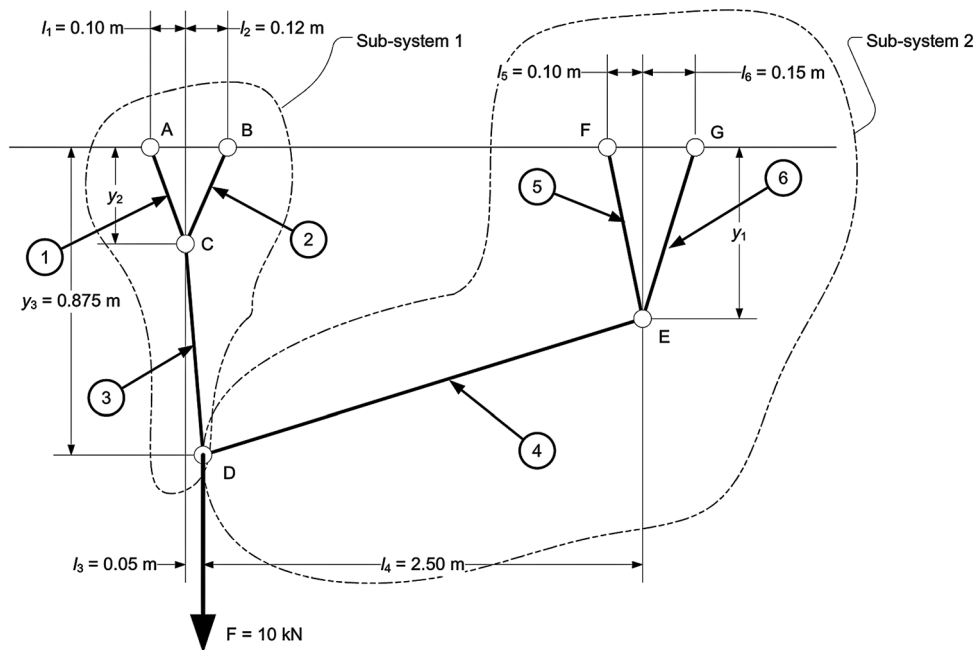


Fig. 9 Structural example configuration

Table 1 Parameter values for structural example

Parameter	Value
E_1, E_2 (GPa)	150
E_3, E_4 (GPa)	180
E_5, E_6 (GPa)	120
$\sigma_{1\max}, \sigma_{2\max}$ (MPa)	100
$\sigma_{3\max}, \sigma_{4\max}$ (MPa)	120
$\sigma_{5\max}, \sigma_{6\max}$ (MPa)	80
$\delta_{D\max}$ (mm)	2.50

the iteration steps. Γ_1 is very small compared to Γ_2 at the initial iterations, so suspending x_1 has little effect on the iteration process. Changes in the scaling of the variables $x_1, x_2,$ and x_3 would change the sensitivities of $f_1, f_2,$ and f_3 with respect to the design variables. However, in the calculation of Γ , this will be cancelled out by the necessary scaling of the local objective function to achieve the same optimal solution for the overall system.

The HCS strategy demonstrated considerable computational efficiency for the model coordination method by suspending system 1. The supersystem problem required 60 iterations and the solution of 180 system optimization problems. With system 1 suspended, the problem required 60 iterations but only 120 system optimization solutions. The savings gained by ignoring system 1 for a limited time are larger than the computational burden of solving Eq. (26).

6 Optimization of a Simple Structure Using Suspension Strategy

Consider a simple structure, as shown in Fig. 9. A load, F , is applied at point D. The structure is to be optimized for minimum material usage, subject to constraints on stress and the displacement at point D, as given in Eqs. (32)–(36). The design variables are the cross-sectional area of each bar, A_1 – A_6 , and the vertical dimensions y_1 and y_2 . Parameter values for l_1 – $l_6, y_3,$ and F are given in Fig. 9, with all dimensions in meters. Maximum stress, deflection, and elastic modulus values are given in Table 1.

$$\begin{aligned} \min_{A_i, i=1, \dots, 6, y_1, y_2} & A_1 \sqrt{l_1^2 + y_2^2} + A_2 \sqrt{l_2^2 + y_2^2} + \\ & A_3 \sqrt{(y_3 - y_2)^2 + l_3^2} + A_4 \sqrt{(y_3 - y_1)^2 + l_4^2} + \\ & A_5 \sqrt{l_5^2 + y_1^2} + A_6 \sqrt{l_6^2 + y_1^2} \end{aligned} \quad (32)$$

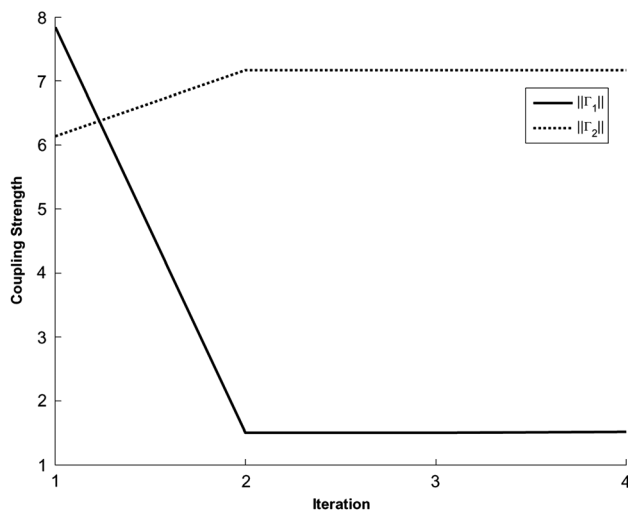


Fig. 10 Comparison of coupling strength in structural example

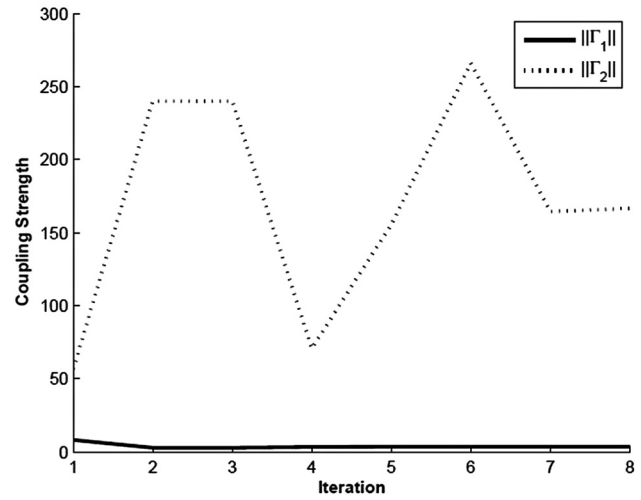


Fig. 11 Comparison of coupling strength in DC motor example

$$\text{subject to } g_i = F_i/A_i - \sigma_{i\max} \leq 0, \quad i = 1, \dots, 6 \quad (33)$$

$$g_7 = \delta_D - \delta_{D\max} \leq 0 \quad (34)$$

$$g_8 = y_1 - y_3 \leq 0 \quad (35)$$

$$g_9 = y_2 - y_3 \leq 0 \quad (36)$$

The structure is partitioned into two subsystems, as shown in Fig. 9, with the subsystem objectives and constraints as follows:

Subsystem 1:

$$\begin{aligned} \min_{A_1, A_2, A_3} & f_1 = A_1 \sqrt{l_1^2 + y_2^2} + A_2 \sqrt{l_2^2 + y_2^2} + \\ & A_3 \sqrt{(y_3 - y_2)^2 + l_3^2} \end{aligned} \quad (37)$$

$$\text{subject to } g_i = F_i/A_i - \sigma_{i\max} \leq 0, \quad i = 1, \dots, 3 \quad (38)$$

Subsystem 2:

$$\begin{aligned} \min_{A_4, A_5, A_6} & f_2 = A_4 \sqrt{(y_3 - y_1)^2 + l_4^2} + A_5 \sqrt{l_5^2 + y_1^2} + \\ & A_6 \sqrt{l_6^2 + y_1^2} \end{aligned} \quad (39)$$

$$\text{subject to } g_i = F_i/A_i - \sigma_{i\max} \leq 0, \quad i = 4, \dots, 6 \quad (40)$$

Subsystem 3:

$$\begin{aligned} \min_{y_1, y_2} & A_1 \sqrt{l_1^2 + y_2^2} + A_2 \sqrt{l_2^2 + y_2^2} + \\ & A_3 \sqrt{(y_3 - y_2)^2 + l_3^2} + A_4 \sqrt{(y_3 - y_1)^2 + l_4^2} + \\ & A_5 \sqrt{l_5^2 + y_1^2} + A_6 \sqrt{l_6^2 + y_1^2} \end{aligned} \quad (41)$$

Table 2 Values of parameters for DC motor

Parameter	Value
ρ_{Cu} (kg/m ³)	8890
ρ_{Fe} (kg/m ³)	7750
A_{wa} (mm ²)	2
A_{wf} (mm ²)	0.5
L_{wa} (cm)	50
L_{wf} (cm)	500

Table 3 Optimal values of design variables for DC motor

Variable	Value
D (cm)	6.9
d_s (cm)	0.69
L (cm)	8.3
K_p	0.81
K_i	5.6
K_d	0.00062
n_d (rad/s)	99.8
V_d (V)	30.0

$$\text{subject to } g_7 = \delta_D - \delta_{D_{\max}} \leq 0 \quad (42)$$

$$g_8 = y_1 - y_3 \leq 0 \quad (43)$$

$$g_9 = y_2 - y_3 \leq 0 \quad (44)$$

A suspension strategy was implemented in this problem, with a starting point of $A_1 = 500 \text{ mm}^2$, $A_2 = 25 \text{ mm}^2$, $A_3 = 1500 \text{ mm}^2$, $A_4 = 50 \text{ mm}^2$, $A_5 = 100 \text{ mm}^2$, $A_6 = 100 \text{ mm}^2$, $y_1 = 0.125 \text{ m}$, and $y_2 = 0.25 \text{ m}$. Subsystem 1 was suspended when the magnitude of Γ_1 was less than one fourth that of Γ_2 . In this case, the system converged to an optimal solution in four iterations, with results of $A_1 = 682 \text{ mm}^2$, $A_2 = 38 \text{ mm}^2$, $A_3 = 826 \text{ mm}^2$, $A_4 = 69 \text{ mm}^2$, $A_5 = 200 \text{ mm}^2$, $A_6 = 220 \text{ mm}^2$, $y_1 = 0.50 \text{ m}$, and $y_2 = 0.27 \text{ m}$. In three of those four iterations, subsystem 1 was suspended. By comparison, when the system was optimized without a suspension strategy, seven iterations were required to reach the same optimal solution. Thus, the suspension strategy resulted in substantial computational savings, since it both required fewer iterations and optimized only one of the two subsystems in most iterations. This result is due to the weakness of the coupling of subsystem 1, as shown in Fig. 10.

7 Optimization of a DC Motor Using Suspension Strategy

In this example, a DC motor with a proportional-integral-derivative (PID) controller is optimized using the model coordination strategy described previously. The model used for the optimization was developed by Reyer and Papalambros [28]. The subsystems consist of the motor design and motor control. The objective functions for these subsystems are given by Eqs. (45) and (46), respectively; the motor design is to be optimized for minimum weight, and the motor control is to be optimized for the minimum value of a quadratic cost function. This cost function includes both the error in the motor speed and the maximum voltage requirement. The overall system objective is a weighted sum of the two individual objectives, with differing weights assigned to the design and control. The constraints for the subsystems are given by Eqs. (1) and (2) in Ref. [28], respectively.

$$f_1(D, L, d_s) = W \quad (45)$$

$$= \rho_{Cu}(A_{wa}L_{wa} + A_{wf}L_{wf}) + \rho_{Fe}L\pi(D + d_s)^2$$

$$f_2(K_p, K_i, K_d) = J = \int_0^{t_f} \omega_{\text{err}}^T \mathbf{Q} \omega_{\text{err}} dt + V_{\max} \quad (46)$$

$$f_3(n_d, V_d) = 0.1W + 0.9J \quad (47)$$

In subsystem 1, the design variables are the rotor diameter, D , depth of slots, d_s , and rotor axial length, L . In subsystem 2, the design variables are the gains of the PID controller, K_p , K_i , and K_d . In the overall system, the design variables are the design speed of the motor, n_d , and the design voltage, V_d . The starting point for

the motor design was set at $D = 7 \text{ cm}$, $d_s = 0.8 \text{ cm}$, and $L = 15 \text{ cm}$, and the starting point for the control optimization was set at $K_p = 0.9$, $K_i = 5.0$, and $K_d = 0.01$. The values of parameters are given in Tables 2 and 3. The system was optimized both with and without suspension. In both cases, the system optimization converged to the optimal values given in Table 3. In the case where no suspension strategy was employed, the system design was complete in 16 iterations. However, when a suspension strategy was used, 8 iterations were required. In addition to requiring fewer iterations, the majority of the iterations required less computation. At each step, the coupling values were computed and compared, and the coupling strength for sub-system 1 was substantially less than that for sub-system 2, as shown in Fig. 11. Therefore, sub-system 1 was suspended throughout the optimization and re-connected at the final iteration. The calculation of the coupling values did add to the computational time for a full iteration between subsystems, but it reduced the number of iterations and the function evaluations required in each iteration. The decomposed optimization with suspension strategy required 4.1 minutes to run, in contrast to a run time of 30 minutes for the decomposed optimization with no suspension. In summary, in this problem the suspension strategy offers reduction of iterations by half and of computational time by a factor of seven. Such gains cannot be claimed to be universal and will vary with the specific problem structure.

8 Conclusion

This article introduced a coupling strength measure for a general nonhierarchical decomposed design optimization problem. The coupling strength measure accounts for optimality by including the optimality conditions of the decomposed supersystem along with the analysis equations in a modified form of the global sensitivity equations.

Numerical computation of the coupling function involves solving a set of linear equations that requires first and second order derivative information of the objectives, constraints, and analysis equations. First-order information can be readily available from the individual system optimization problem, but obtaining second order information can be very expensive, as shown in the optimization of a DC motor. Future work must consider methods to deal with this cost. Such methods may include techniques for efficient estimation of coupling, or the use of *a priori* coupling determination methods to identify problems in which the computational expense of computing the necessary derivatives will be justified. In addition, the coupling function depends on approximate Lagrange multipliers computed under the assumption that activity does not change. Some sort of active set strategy must be introduced to address the activity assumptions. It may also be beneficial to consider individual values within the vector Γ , in order to suspend particular variables within a given subsystem.

The Hierarchical Coupling Suspension strategy has been shown to be promising in conjunction with the model coordination method. Future work may explore HCS for other MDO and multilevel algorithms, such as collaborative optimization [29], and the use of suspension strategies in analytical target cascading [30]. Substantial numerical testing remains to be done for problems with increased complexity as well as with high function evaluation costs. The impact of problem scaling must be also investigated, particularly in problems where the subsystem objectives are not commensurate and their respective weights are based on engineering intuition. The means to determine the boundary between “weak” and “strong” coupling is also a critical consideration, currently based on engineering intuition, which merits further investigation. Isolating system elements that have weak coupling to the overall system from system redesign iterations is common in practice. This paper offers a more rigorous implementation of this practice and showed that significant computational advantage may be gained for some problems.

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Nomenclature

A_i = cross-sectional area of Bar # i , $i = 1, \dots, 6$
 A_{wa} = cross-sectional area of armature wire for DC motor
 A_{wf} = cross-sectional area of field wire for DC motor
 D = diameter of rotor diameter for DC motor
 d_s = slot depth for DC motor
 E_i = elastic modulus in Bar # i , $i = 1, \dots, 6$
 F_i = force present in Bar # i , $i = 1, \dots, 6$
 f_i = objective function associated with system i ,
 $f_i : \mathbb{R}^{q_i} \rightarrow \mathbb{R}$
 $\partial f / \partial \mathbf{x}$ = gradient vector of $f(\mathbf{x})$ —a row vector
 F = objective function representing the supersystem
 objective, $F : \mathbb{R}^{N+} \sum_{i=1}^N n_i \rightarrow \mathbb{R}$
 \mathbf{g}_i = inequality constraints associated with system i ,
 $\mathbf{g}_i : \mathbb{R}^{q_i} \rightarrow \mathbb{R}^{m_i}$
 $\partial \mathbf{g} / \partial \mathbf{x}$ = Jacobian matrix of \mathbf{g} with respect to \mathbf{x} ; it is $m \times n$, if \mathbf{g} is an m -vector and \mathbf{x} is an n -vector
 \mathbf{h}_i = equality constraints associated with system i ,
 $\mathbf{h}_i : \mathbb{R}^{q_i} \rightarrow \mathbb{R}^{o_i}$
 J = cost function for control of DC motor
 k = (subscript only) denotes values at k th iteration
 K_d = derivative gain for DC motor
 K_i = Integral gain for DC motor
 K_p = proportional gain for DC motor
 l_i = horizontal span of Bar # i , $i = 1, \dots, 6$
 l_{ij} = number of interaction variables associated with system interaction variable y_{ij}
 L = rotor axial length for DC motor
 L_{wa} = length of armature wire for DC motor
 L_{wf} = length of field wire for DC motor
 m_i = number of inequality constraints associated with system inequality constraint \mathbf{g}_i
 n_d = design speed of DC motor
 n_i = number of design variables associated with system design variable \mathbf{x}_i
 N = total number of systems
 o_i = number of equality constraints associated with system equality constraint \mathbf{h}_i
 q_i = total number of design and interaction variables associated with system i , $q_i \triangleq \sum_{j=1}^N (n_j + l_{ij})$
 $d\hat{\mathbf{x}}_j / d\mathbf{x}_i$ = gradient of optimal solution of system j with respect to \mathbf{x}_i for the optimization problem with \mathbf{x}_i suspended
 \mathbb{R}^n = n -dimensional Euclidean (real) space
 V_d = design voltage of DC motor
 V_{\max} = maximum voltage requirement for DC motor
 W = weight of DC motor
 \mathbf{x}_i = vector of design variables associated with system i ,
 $\mathbf{x}_i \in \mathbb{R}^{n_i}$
 y_1, y_2, y_3 = vertical span of links in structural example
 \mathbf{y}_{ij} = data transfer or interaction variable vector from system i to system j where $\mathbf{y}_{ij} \in \mathbb{R}^{l_{ij}}$; $\mathbf{y}_{ii} \in \mathbb{R}^{l_{ii}}$ from system i to itself represents system simulation (analysis) models
 $\delta_{D_{\max}}$ = maximum allowable deflection at point D
 Γ_i = optimization coupling function vector associated with system design variable \mathbf{x}_i
 ρ_{Cu} = density of copper wire in DC motor
 ρ_{Fe} = density of iron core of DC motor
 $\delta_{D_{\max}}$ = maximum allowable stress in Bar # i , $i = 1, \dots, 6$
 ω_{err} = error in speed of DC motor
 $+, \times, \|\cdot\|$ = matrix sum, matrix product, and Euclidean norm, respectively
 \triangleq = definition

References

- [1] Haftka, R., Martinovic, Z., Hallauer, W., Jr., and Schamel, G., 1986, "An Analytical and Experimental Study of a Control System's Sensitivity to Structural Modifications," *AIAA J.*, **25**(2), pp. 310–315.
- [2] Bloebaum, C. L., English, K., and Miller, E., 2001, "Development of Multiple Cycle Coupling Suspension in the Optimization of Complex Systems," *Struct. Multidiscip. Optim.*, **22**(4), pp. 268–283.
- [3] Fathy, H. K., Reyer, J. A., Papalambros, P. Y., and Ulsoy, A. G., 2001, "On the Coupling Between the Plant and Controller Optimization Problems," in *Proceedings of the American Control Conference*, ASME, pp. 1864–1869.
- [4] Steward, D., 1981, *Systems Analysis and Management: Structure, Strategy and Design*, Petrocelli Books, New York.
- [5] Gebala, D., and Eppinger, S., 1991, "Methods for Analyzing Design Procedures," in 3rd International Conference on Design Theory and Methodology.
- [6] Wagner, T. C., 1993, "A General Decomposition Methodology for Optimal System Design," Ph.D. thesis, University of Michigan, Ann Arbor, MI.
- [7] Wagner, T. C., and Papalambros, P. Y., 1993, "Implementation of Decomposition Analysis in Optimal Design," in *Proceedings of the 19th Annual ASME Design Automation Conference*, pp. 327–335.
- [8] Krishnamachari, R. S., and Papalambros, P. Y., 1997, "Hierarchical Decomposition Synthesis in Optimal Systems Design," *ASME J. Mech. Des.*, **119**(4), pp. 448–457.
- [9] Michelena, N. F., and Papalambros, P. Y., 1995, "Optimal Model-Based Decomposition of Powertrain System Design," *ASME J. Mech. Des.*, **117**(4), pp. 499–505.
- [10] Sobieszcwanski-Sobieski, J., 1990, "Sensitivity of Complex, Internally Coupled Systems," *AIAA J.*, **28**(1), pp. 153–160.
- [11] English, K., 2001, "Combined System Reduction and Sequencing in Complex System Optimization," AIAA Paper No. 2002-5412, in *9th AIAA/ISSMO Symposium on Multidisciplinary Analysis and Optimization*, Vol. 22, pp. 268–283.
- [12] Sobieszcwanski-Sobieski, J., Agte, J., and Sandusky, R., 2000, "Bilevel Integrated System Synthesis," *AIAA J.*, **38**(1), pp. 164–172.
- [13] Onoda, J., and Haftka, R., 1987, "An Approach to Structure/Control Simultaneous Optimization for Large Flexible Spacecraft," *AIAA J.*, **25**(8), pp. 1133–1138.
- [14] O'Neal, G. P., Min, B.-K., Pasek, Z. J., and Koren, Y., 2001, "Integrated Structural/Control Design of Micro-Positioner for Boring Bar Tool Insert," *J. Intell. Mater. Syst. Struct.*, **12**(9), pp. 617–627.
- [15] Reyer, J. A., Fathy, H. K., Papalambros, P. Y., and Ulsoy, A. G., 2001, "Comparison of Combined Embodiment Design and Control Optimization Strategies Using Optimality Conditions," in *Proceedings of the ASME Design Engineering Technical Conferences*, ASME, Paper No. DETC2001/DAC-21119, pp. 1023–1032.
- [16] Peters, D. L., Papalambros, P. Y., and Ulsoy, A. G., 2009, "On Measures of Coupling Between the Artifact and Controller Optimal Design Problems," in *Proceedings of the ASME Design Engineering Technical Conference & Computers in Engineering Conference*, ASME, Paper No. DETC 2009-86868.
- [17] Peters, D. L., 2010, "Coupling and Controllability in Optimal Design and Control," Ph.D. thesis, University of Michigan, Ann Arbor, MI.
- [18] Peters, D. L., Papalambros, P. Y., and Ulsoy, A. G., 2010, "Relationship Between Coupling and the Controllability Grammian in Co-Design Problems," in *Proceedings of the American Control Conference*.
- [19] Allison, J. T., Kokkolaras, M., and Papalambros, P. Y., 2007, "On Selecting Single-Level Formulations for Complex System Design Optimization," *ASME J. Mech. Des.*, **129**(9), pp. 898–906.
- [20] Allison, J. T., Kokkolaras, M., and Papalambros, P. Y., 2009, "Optimal Partitioning and Coordination Decisions in Decomposition-Based Design Optimization," *ASME J. Mech. Des.*, **131**(8), pp. 1–8.
- [21] Han, J., and Papalambros, P. Y., 2007, "A Sequential Linear Programming Coordination Algorithm for Analytical Target Cascading," in *Proceedings of the ASME 2007 International Design Engineering Technical Conferences & Computers and Information in Engineering Conference*, ASME, Paper No. DETC2007-35361.
- [22] Han, J., and Papalambros, P. Y., 2007, "An slp Filter Algorithm for Probabilistic Analytical Target Cascading," in *Proceedings of the 7th World Congress on Structural and Multidisciplinary Optimization*, pp. 1329–1338.
- [23] Gu, X., Renaud, J. E., and Penninger, C. L., 2006, "Implicit Uncertainty Propagation for Robust Collaborative Optimization," *ASME J. Mech. Des.*, **128**(4), pp. 1001–1013.
- [24] Kirsch, U., 1981, *Optimum Structural Design: Concepts, Methods, and Applications*, McGraw-Hill, New York.
- [25] Papalambros, P., and Wilde, D., 2000, *Principles of Optimal Design*, Cambridge University, Cambridge, UK.
- [26] Sobieszcwanski-Sobieski, J., Barthelemy, J., and Riley, K., 1982, "Sensitivity of Optimum Solutions of Problem Parameters," *AIAA J.*, **20**(9), pp. 1291–1299.
- [27] Schoeffler, J., 1971, *Static Multilevel Systems*, McGraw-Hill, New York.
- [28] Reyer, J. A., and Papalambros, P. Y., 2002, "Combined Optimal Design and Control With Application to an Electric DC Motor," *ASME J. Mech. Des.*, **124**(2), pp. 183–191.
- [29] Braun, R. D., 1996, "Collaborative Optimization: An Architecture for Large-Scale Distributed Design," Ph.D. thesis, Stanford University, Palo Alto, CA.
- [30] Kim, H., 2001, "Target Cascading in Optimal System Design," Ph.D. thesis, University of Michigan, Ann Arbor, MI.